

iceAccretionFoam: Rime and Glaze Ice Accretion

SAE AC-9C Oct 2025 @ CIRA
Capua – Italy

Dr. G. A. Lima da Silva



All skills in a
single company

18 years of experience
applying engineering and
CFD in the industry



Engineering Solutions

Associate Engineer throughout
the development process



Simulation, Test &
Certification

Multi-disciplinary team

Fidelity and long-term partnership

Experience-ready for complex projects

ATS Expertise Areas for Aerospace



Ice Protection

- System Performance
- Ice Accretion and Shapes
- Performance Degradation
- Escaping conditions
- Atmospheric Studies
- Air Data Probes Ice Protection
- Engine Ice Ingestion



Environmental Control Systems ECS

- Air Condition Distribution and Control
- Thermal Comfort
- Avionics Cooling and Dedicated Systems
- Pneumatic System and Ducting
- Smoke Evacuation, Prevention, and Detection
- Oxygen Systems and Pressurization
- Battery Thermal Management System



System Engineering

- System and component failure Analysis
- Design and Development
- Certification Requirements
- Operational Difficulties
- Tests
- Reports at Certification Level



Stress Analysis

- Conceptual design
- ASME, ASTM and ABNT Standards
- Analysis of Reactions and Displacements Thermal
- Expansion
- Hyper elasticity. Fluids and Nonlinearity
- Vibrations

Clients



iceAccretionFoam Project

It is a collaboration between ATS-Aerothermal solutions and Dr. Jasak, original foam developer, Cambridge Univ.



Work Sharing

- **Dr. Jasak:** Initial Code Skeleton, Main Existent foam-ext Core, Solving Issues
- **Dr. Lima da Silva:** Physics, Equations implementation, Coding, Literature Research, Supervising
- **Teixeira da Silva:** Coding to Extend and Improve foam-ext Functions and implement key parts of the iceAccretionFoam, Interface Cambridge-ATS
- **Carvalho:** Support Coding and Validation, Operation and Run of Test Cases

Participation of Dr. Jasak

- Instructions on using **finite areas for simulating liquid water** rivulets and films on highly curved surfaces. Dr. Jasak's unique expertise enabled transferring his foam-ext finite-area code to official OpenFoam.
- Guidance on **immersed boundaries with ice-driven growth**, only available in foam-ext by Dr. Jasak.
- Supervising ATS developers for implementing advanced code parts, like **finite-area interfaces with immersed boundaries**, not ready in the base code. External flow immersed boundary values must be mapped to finite area and back for ice growth simulation.
- Dr. Jasak built preliminary code skeletons, but **ATS implemented the final complete version**. Collaboration focused on meetings for guidance on foam-ext classes and objects.
- His participation was vital; the work involves **core foam-ext modifications** requiring unique, non-market knowledge not available or standard OpenFoam training.

Code Development

- Using GitHub as a **git repository** and version control system, we can manage our code effectively and safely.
- When we make significant changes to a branch, we validate and test them before merging them into the main branch.
- On the other hand, for minor changes, we implement and test them in the main branch.
- Additionally, we can create test versions to demonstrate concepts before merging them into the main branch.
- Employ the repository as **a secure backup** for versions that have proven successful. You can select which version is acceptable or not to safely continue the development process.
- Accessible **to everyone in the project** but it is not public for all

ATS gitHub



ATS-aerothermal-solutions

Q

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Overview

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Projects

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ATS-aerothermal-solutions

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Popular repositories

dataAndModelsCovid19

Public

Forked from [gasilva/dataAndModelsCovid19](#)

Python code to analyze data and predict Covid-19 infection

Jupyter Notebook

jsbsimNew

Public

Forked from [gasilva/jsbsimNew](#)

An open source flight dynamics & control software library

C++

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Updated 4 days ago



iceAccretionFoam_dev

Private

A private repository to safely store and share between the collaborators, the iceAccretionFoam's code/doc/related things

Python

1

0

2

0

Updated on Feb 4



People



Invite someone

Top languages

Physical Model

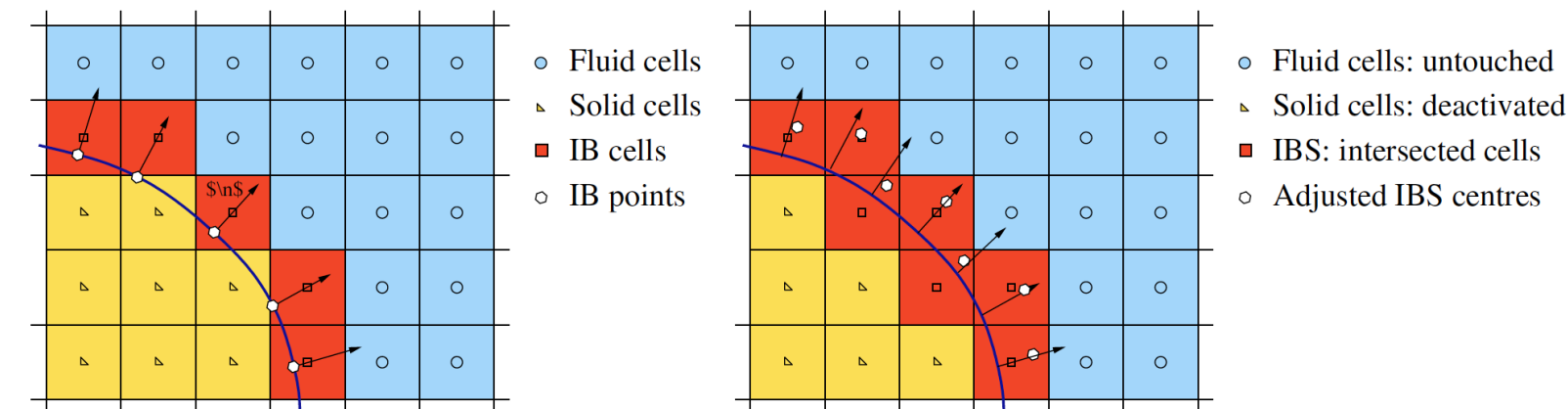


$$\frac{\partial \rho \delta}{\partial t} + \nabla_s \cdot [\rho \delta \mathbf{U}] = S_{\rho \delta}$$
$$\frac{\partial \rho \delta \mathbf{U}}{\partial t} + \nabla_s \cdot [\rho \delta \mathbf{U} \mathbf{U}] = \underbrace{-\delta \nabla_s p}_{\text{pressure based}} + \underbrace{S_{\rho \delta \mathbf{U}}}_{\text{stress based}}$$
$$p = p_{\sigma} + p_{\delta} + p_g$$

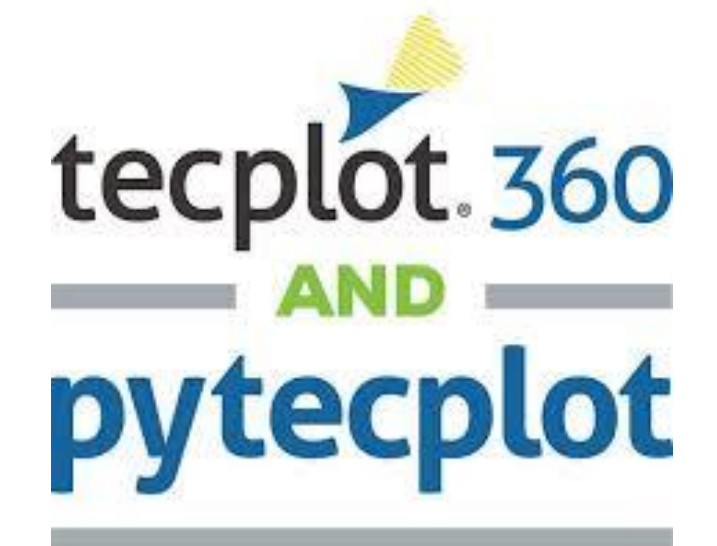
The pressure term, p , is comprised of forces in the wall-normal direction and consists of capillary effects (p_{σ}), hydrostatic pressure head (p_{δ}), and local gas-phase pressure (p_g).

$$S_{\rho \delta \mathbf{U}} = \tau_g + \tau_w + \rho g_t \delta + \tau_{\theta}$$

The stress term, $S_{\rho \delta \mathbf{U}}$, includes the viscous shear stresses (τ_g, τ_w), gravity body force ($\rho g_t \delta$), and stress related to contact-angle force (τ_{θ}).



ENNOVA
TECHNOLOGIES



CFD 3D

Water Film Model
Immersed Boundary

Heat Transfer
and Freezing

Results

Why Foam-Ext?

- Recognized CFD 3D Software (not new and not from scratch)
- Finite Area to simulate runback water hydrodynamics and thermal
- Immersed Boundary to simulate ice growth without mesh deformation or local/global re-meshing
- Turbulent Rough Heat Transfer implemented in OpenFoam of a new wall function specific for icing surfaces based on traditional modeling (Lewice/Stanford)
- Implementations of Laminar-Turbulent Transition available in literature for OpenFoam/foam-ext triggered by roughness size
- driftEulerFoam already implemented for Eulerian droplets trajectories, including clouds and distributions
- Possibility to run in the cloud through web interface - SAAS

Compressible Full 3D Solver with Eulerian Phase

Immersed Boundary (shape changes)

Finite Area Method (liquid water flow)

Rough Turbulent Convective Heat Transfer Wall Function

Laminar-Turbulent Transition on Rough Surface

2D Shell Thermal Conduction Heat Transfer

iceAccretionFoam implemented as a solver of foam-ext 5.0

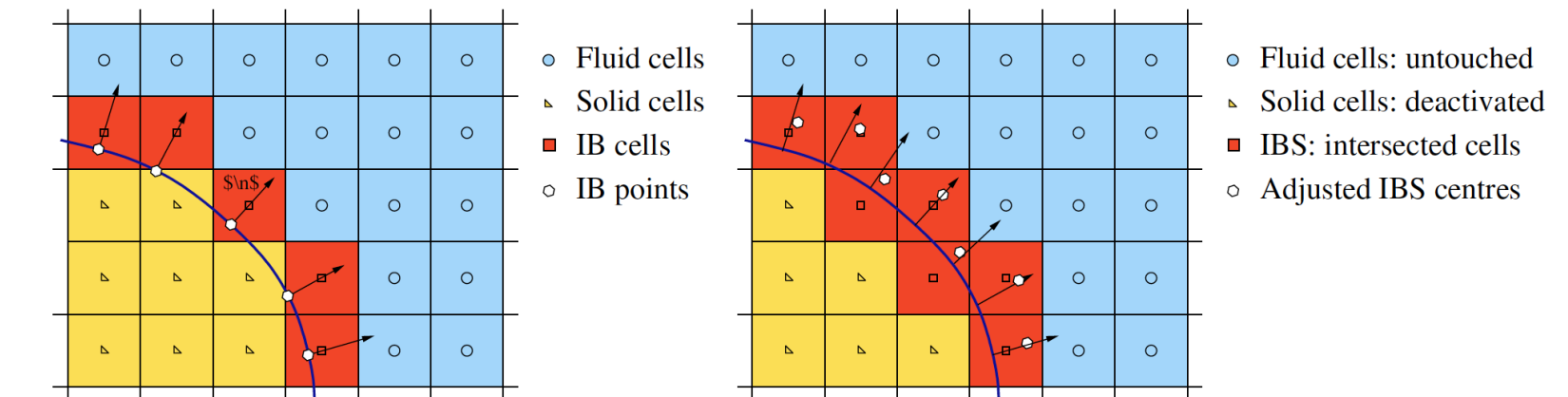


Figure 2: Traditional IBM and New Implementation of IBS after Jasak (23)

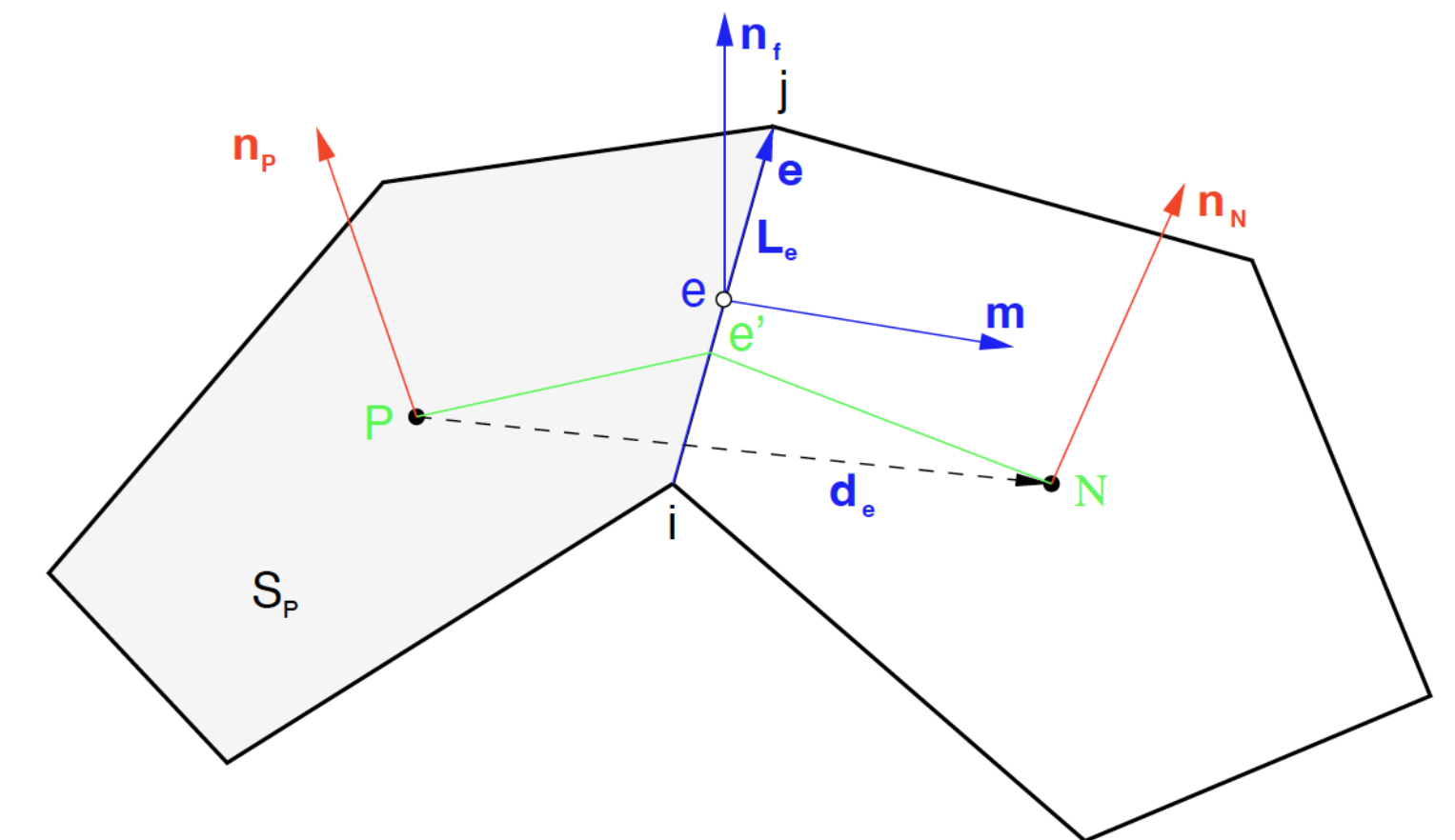
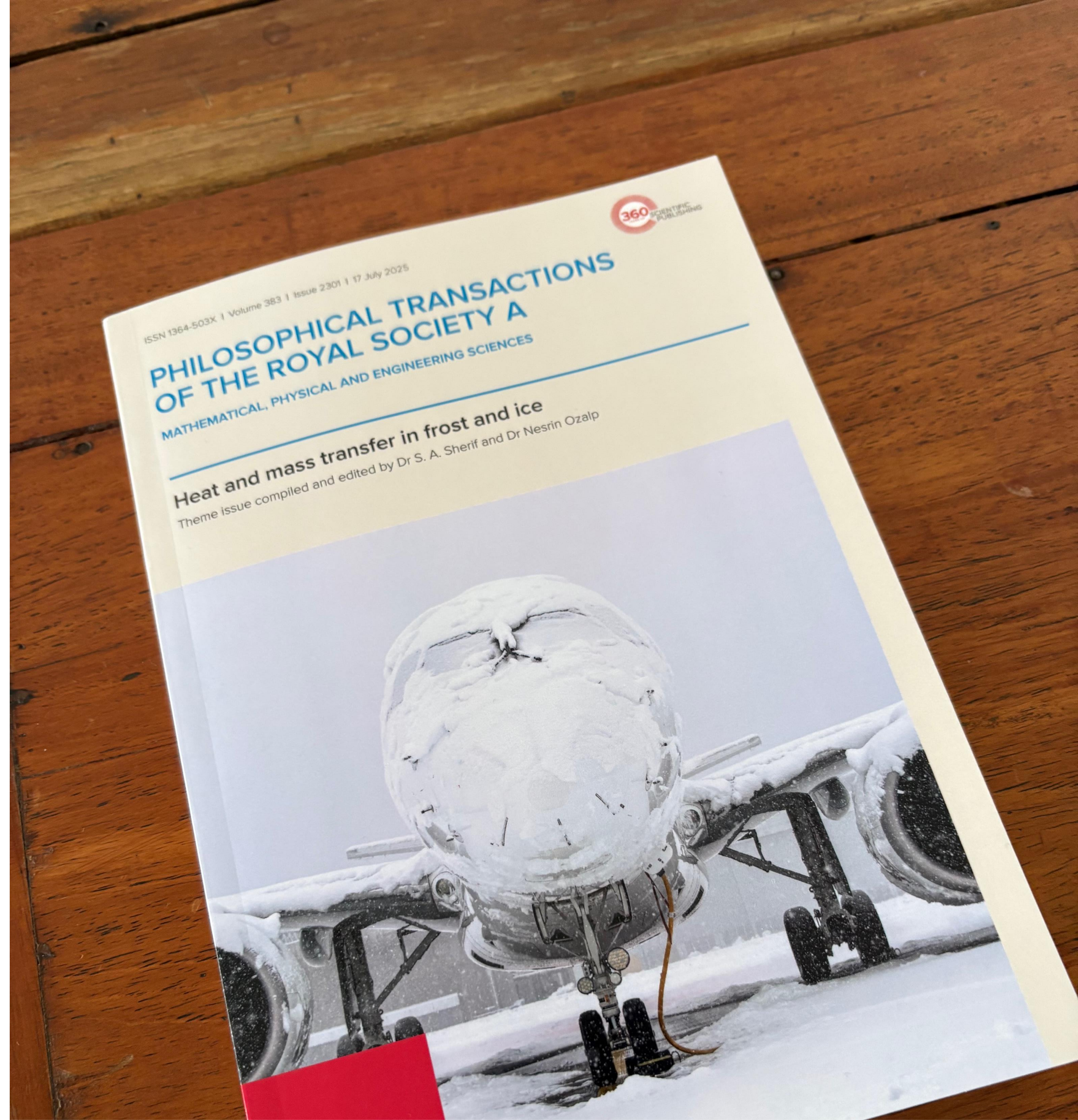


Figure 1: Face-edge interpolation in edge based local coordinate system after Jasak (18)

iceAccretionFoam Publications

Three papers published:

- Carvalho et al. (2025). Advancing the iceAccretionFoam Solver: Glaze Ice Accretion. **Phil. Trans. R. Soc.** DOI: 10.1098/rsta.2024.0361
- Teixeira da Silva et al. (2025). Introducing the iceAccretionFoam Solver: Impingement and Rime Ice. **Phil. Trans. R. Soc.** DOI: 10.1098/rsta.2024.0362
- Teixeira da Silva et al. Initial Development of a CFD Icing Tool. **ICAS 2024-1112**



iceAccretionFoam Code

Current Scope

- A transient solution of the flow by using Unsteady Reynolds Averaged Navier-Stokes compressible equations (URANS) in a multi-step approach;
- Compressible flow energy equation with kinetic heating;
- Fully turbulent momentum and heat transfer over a sand-grain rough surface
- An immersed boundary method that updates the frontier of the solid surface;
- A finite-area method that simulate the water hydrodynamics on the airfoil surface;
- A Eulerian-dispersed phase model with driftEulerFoam solver for compressible flow.

Main Physical Assumptions

- 1) All impinging supercooled liquid water will form a film with $\rightarrow$ no surface tension breakdown
- 2) A normal growth of ice rule is applied based on growth rate of ice in the film;
- 3) A simplistic mathematical freezing fraction is implemented preliminarily;
- 4) Water thermal balance and convective heat and mass transfer are considered;
- 5) Air kinetic heating and pressure effects are calculated in compressible flow;
- 6) Liquid water film exist on the surface of the airfoil;
- 7) The liquid film is solved in the non-deformed solid surface but with effects of pressure gradient, shear, heat transfer, water droplets momentum and mass flux from deformed surface at immersed boundary (external flow over ice shape);

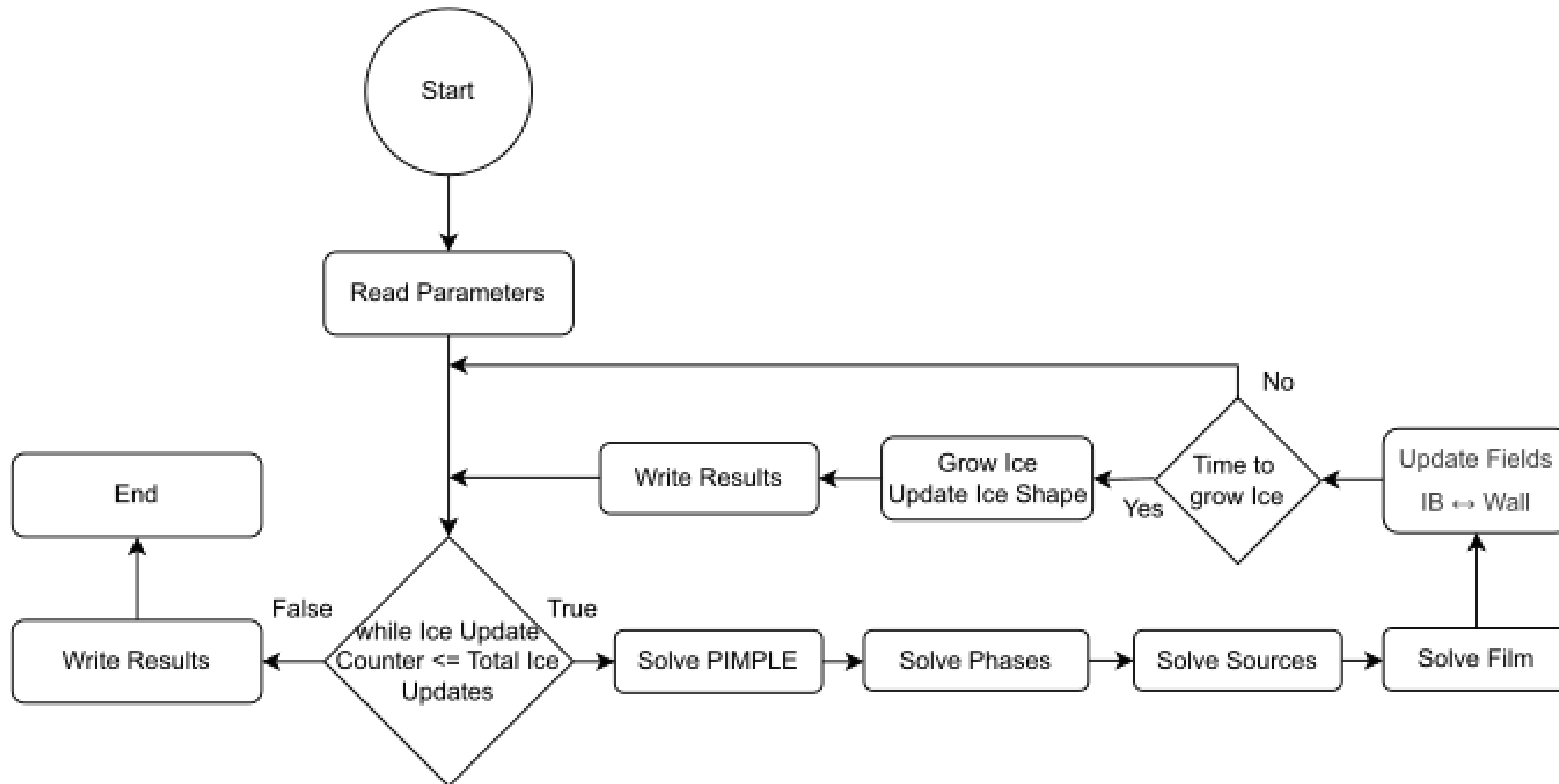
Numerical Model

- 1) rhoPimpleFoam* for main flow compressible URANS with added kinetic heating
- 2) Special wall functions implemented for fully rough surface
- 3) Water mass, momentum and energy conservation for water film
- 4) Eulerian droplets trajectories – U_{phase} and α for each diameter

5) **In each time step a height value is added:**
$$h_{ice,i} = \Delta t_{accretion} \cdot \alpha_i \cdot U_{phase,i} \cdot \frac{\rho_{water}}{\rho_{ice}}$$

6) Lewice Ice density model:
$$\rho_{ice} = 1000 \cdot \exp[0.15 \cdot (1 + 6043 \cdot S^{-2.65})]$$
$$S = \frac{MVD^{0.82} \cdot U_{\infty}^{0.59} \cdot LWC^{0.21}}{d_{le}^{0.48} \cdot (-T_{rec}^{0.23})}$$

Code Implementation



Planned Validation Process

Cp & Beta

2D Rime Ice

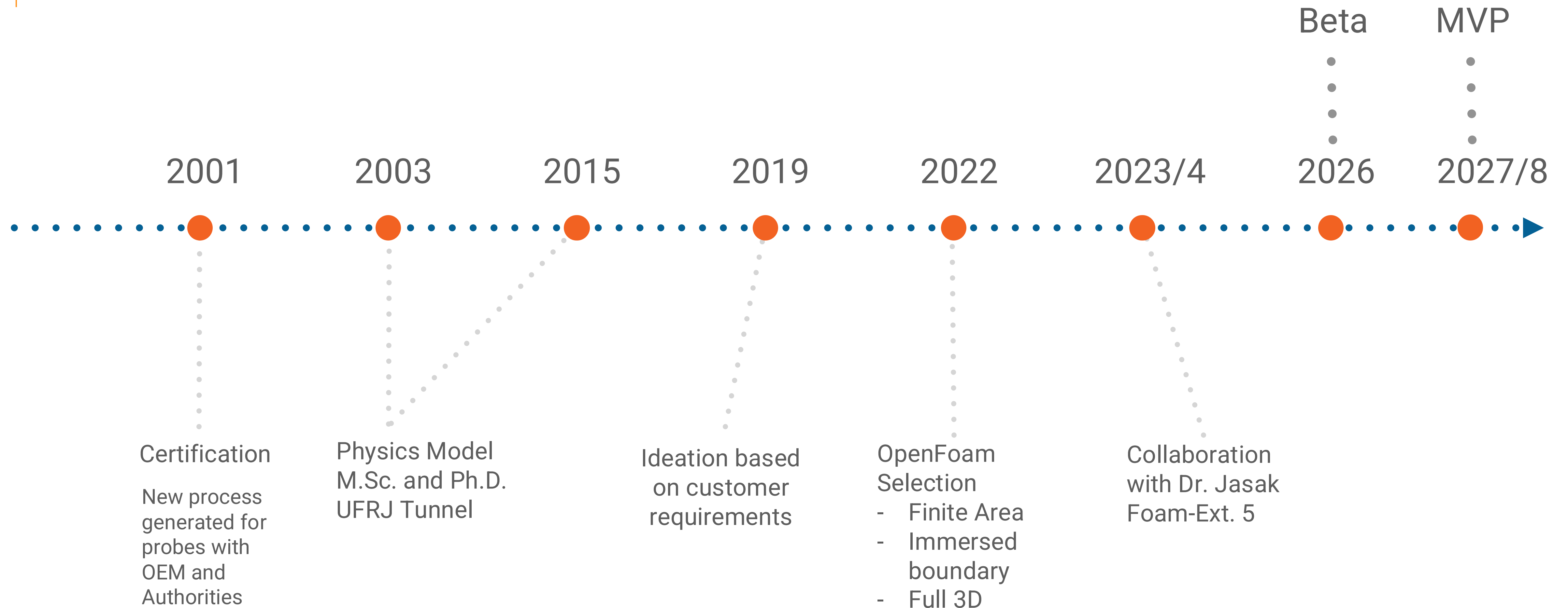
2D Glaze Ice

3D Shapes & SLD

Thermal Ice Protection



Roadmap



Technology is designed to be a step ahead

Definitions

Droplet Distribution

Table 2 – Distribution of Wright [16] with MVD = 21 μm

%LWC	D(I)
0.1390	8.6
0.0958	12.5
0.0997	15.5
0.1220	18.5
0.1208	21.5
0.1115	24.5
0.0917	27.5
0.0946	31.6
0.0899	48.2
0.0350	95.9

Definitions

Pressure Coefficient

$$C_p = \frac{p - p_\infty}{\frac{1}{2}\rho U_\infty^2},$$

$$C_p^* = C_p \cdot \frac{1}{\sqrt{1 - M_\infty^2}}$$

Collection Efficiency

$$\beta_i = \frac{\alpha_i \cdot \vec{U}_{phase,i} \cdot \vec{n}}{\alpha_\infty \cdot |\vec{U}_{phase,\infty}|}$$

$$\beta_{tot} = \sum_{i=1}^n \frac{\alpha_i \cdot \beta_i}{\alpha_\infty}$$

$$\alpha_\infty = \sum_{i=1}^n \alpha_i$$

Freestream α

$$\alpha_\infty = \frac{LWC_\infty}{\rho_{water}}$$

Simplistic Freezing Fraction

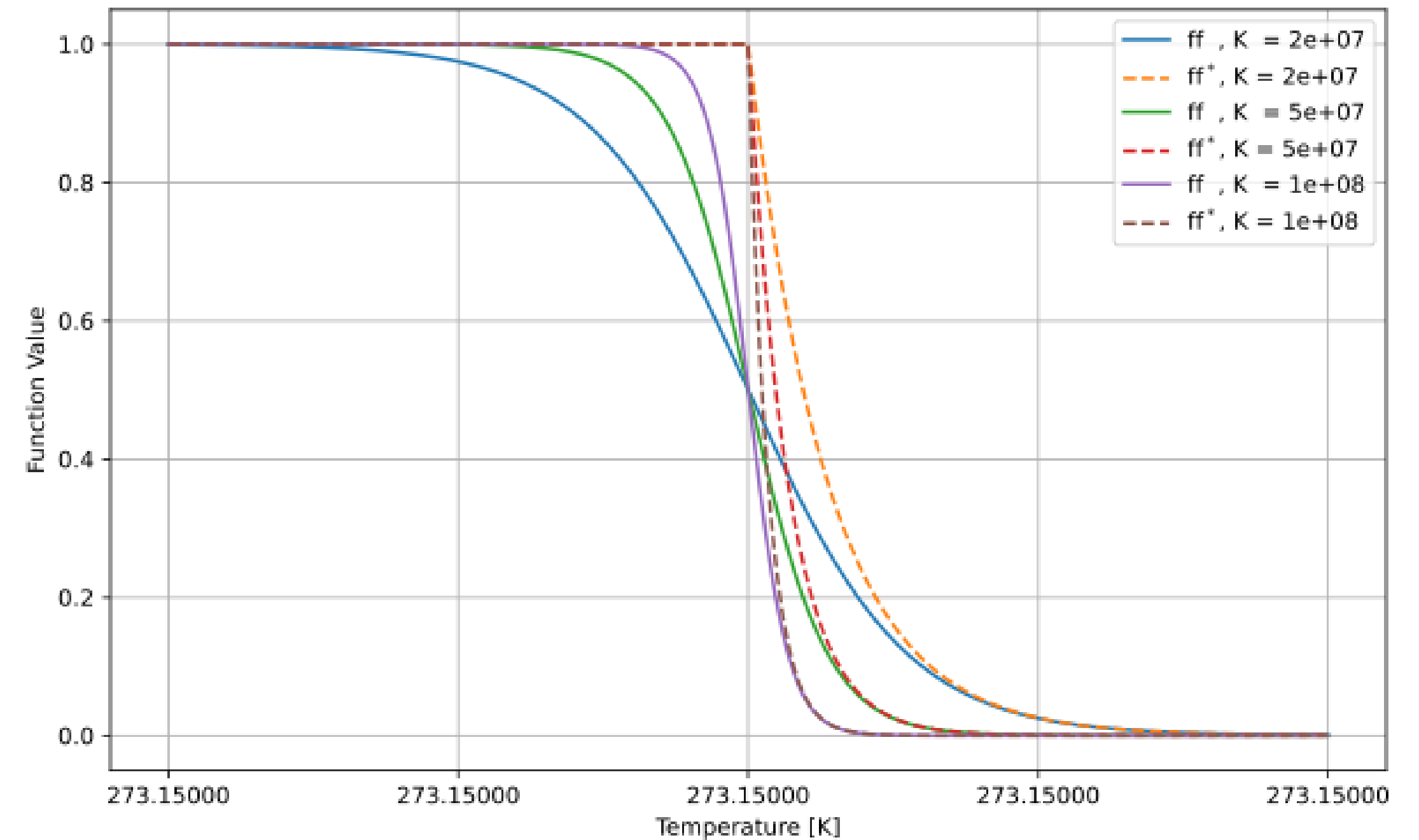
$$T_{diff} = \frac{T_f - 273.15}{273.15}$$

$$ff = 1 - \left(0.5 + 0.5 \tanh \left(K_{Freezing} \cdot T_{diff} \right) \right)$$

$$\frac{\partial m}{\partial t}_{ice,cv} = ff \cdot \frac{\partial m}{\partial t}_{total,cv} = ff^* \cdot \frac{\partial m}{\partial t}_{liquid,cv}$$

$$\frac{\partial m}{\partial t}_{total,cv} = \frac{\partial m}{\partial t}_{liquid,cv} + \frac{\partial m}{\partial t}_{ice,cv}$$

$$\Rightarrow ff^* = \frac{ff}{1 - ff}$$



Freezing and Thermal Model

Water Film Conservation Finite Area Formulation

Mass =
Film Height

$$\frac{\partial h_f}{\partial t} + \nabla \cdot (\phi_s \cdot h_f) = 2 \cdot S_m - S_f - S_e$$

$$S_m = \frac{\sum \alpha_i \cdot \phi_{\text{phase},i}}{|A|}$$

$$S_f = \text{ff}^* \cdot \frac{\partial h_f}{\partial t}_{\text{cv}}$$

$$S_e = \frac{\dot{m}''_{\text{evap}}}{\rho_w},$$

Energy =
Film Height X Temperature

$$\frac{\partial h_f \cdot T_f}{\partial t} + \nabla (\rho_w \cdot C_w \cdot \phi_{2s} \cdot T_f) - k \cdot h_f \cdot \nabla^2 T = h_{\text{conv}} \cdot (T_{\infty} - T_f) + S_{\text{mi}} + S_{\text{Hf}} + S_{\text{He}}$$

$$S_{\text{mi}} = \rho_w \cdot C_w \cdot S_m \cdot T_d$$

$$S_{\text{Hf}} = \text{ff}^* \cdot \frac{\partial h_f}{\partial t} \cdot \rho_w \cdot h_{\text{ls}}$$

$$S_{\text{He}} = (1 - \text{ff}^*) \cdot \dot{m}''_{\text{evap}} \cdot h_{\text{lv}}$$

Momentum =
Film Height X Velocity

$$\frac{\partial h_f \vec{U}_s}{\partial t} + \nabla \cdot (\phi_{2s} \cdot \vec{U}_s) + 0.0125 \cdot C_{f,w} |\vec{U}_s| \cdot \vec{U}_s = (S_{\tau} + S_i + S_g) \cdot \vec{b}_{\parallel} + S_{\text{grad}P} + S_{\text{bl}}$$

$$\phi_{2s} = \phi_s \cdot h_f$$

$$S_{\tau} = \frac{\vec{\tau} \cdot h_f}{\rho_w}$$

$$S_i = \frac{\sum \alpha_i \cdot \phi_{\text{phase},i} \cdot \vec{U}_{\text{phase},i}}{|A|}$$

$$S_g = \vec{g} \cdot h_f$$

$$S_{\text{grad}P} = -\vec{\nabla}(p_s \cdot h_f) \\ = \frac{p_s}{\rho_w} \cdot \vec{\nabla} h_f.$$

Rough Wall Functions Immersed Boundary

$$\Delta B = \begin{cases} 0 & K_s^+ \leq 2.5 \\ \frac{1}{\kappa} \ln \left[\frac{K_s^+ - 2.25}{87.75} + C_s K_s^+ \right] \sin[0.4258 \cdot (\ln K_s^+ - 0.811)] & 2.5 < K_s^+ < 90 \\ \frac{1}{\kappa} \ln(1 + C_s K_s^+) & K_s^+ \geq 90 \end{cases}$$

Stanford
Rough Wall

Where C_s depends on the shape and distribution of the roughness elements and,

$$\mu_t = \mu \left(\frac{y^+ \kappa}{\ln(Ey^+/e^{\kappa\Delta B})} - 1 \right) \quad y^+ = y \frac{\rho C_\mu^{0.25} k^{0.5}}{\mu} \quad \text{and} \quad K_s^+ = K_s \frac{\rho C_\mu^{0.25} k^{0.5}}{\mu}$$

Double Resistance Analogy
Not Reynolds Analogy

$$\frac{u_\tau}{u} = \sqrt{C_f/2} \quad \text{and} \quad u_\tau = \sqrt{\tau/\rho} \quad \text{with} \quad \tau = (\mu + \mu_t) \cdot |\nabla U \cdot \vec{n}|,$$

$$\alpha_t = \frac{\mu_t \cdot \eta \cdot A}{Pr_t} \quad \eta = \frac{1}{Pr_t + \frac{\sqrt{C_f/2}}{St_k}}$$

Heat and Mass Transfer Imersed Boundary

$$T_d = T_\infty + \frac{V_d^2}{2 * C_w}$$

$$\dot{q}''_{\text{wall}} = \alpha_{\text{eff}} \cdot \nabla h = \alpha_{\text{eff}} \cdot c_p \cdot \nabla T$$

$$h_{\text{conv}} = \left| \frac{\dot{q}''_{\text{wall}}}{T_{\text{wall}} - T_\infty} \right|,$$

$$B_m = \frac{\text{mf}_{\text{H}_2\text{O},S} - \text{mf}_{\text{H}_2\text{O},G}}{\text{mf}_{\text{H}_2\text{O},S} - 1},$$

$$\dot{m}''_{\text{evap}} = g_m \cdot B_m$$

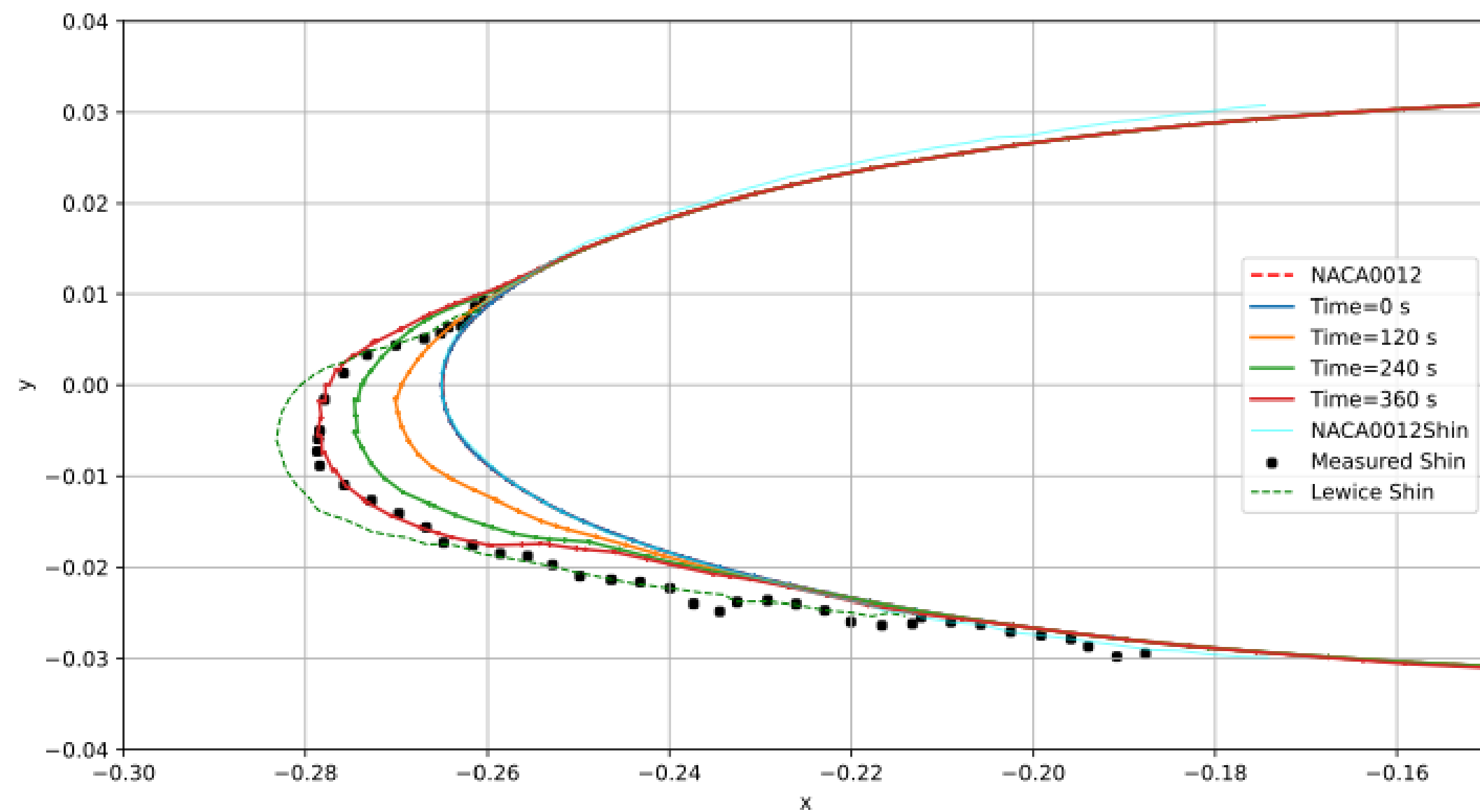
$$g_m = \frac{h_{\text{conv}}}{c_p} \cdot \text{Le}^{2/3} \cdot \frac{\ln(1 + B_m)}{B_m},$$

Ice Growth for Rime and Glaze

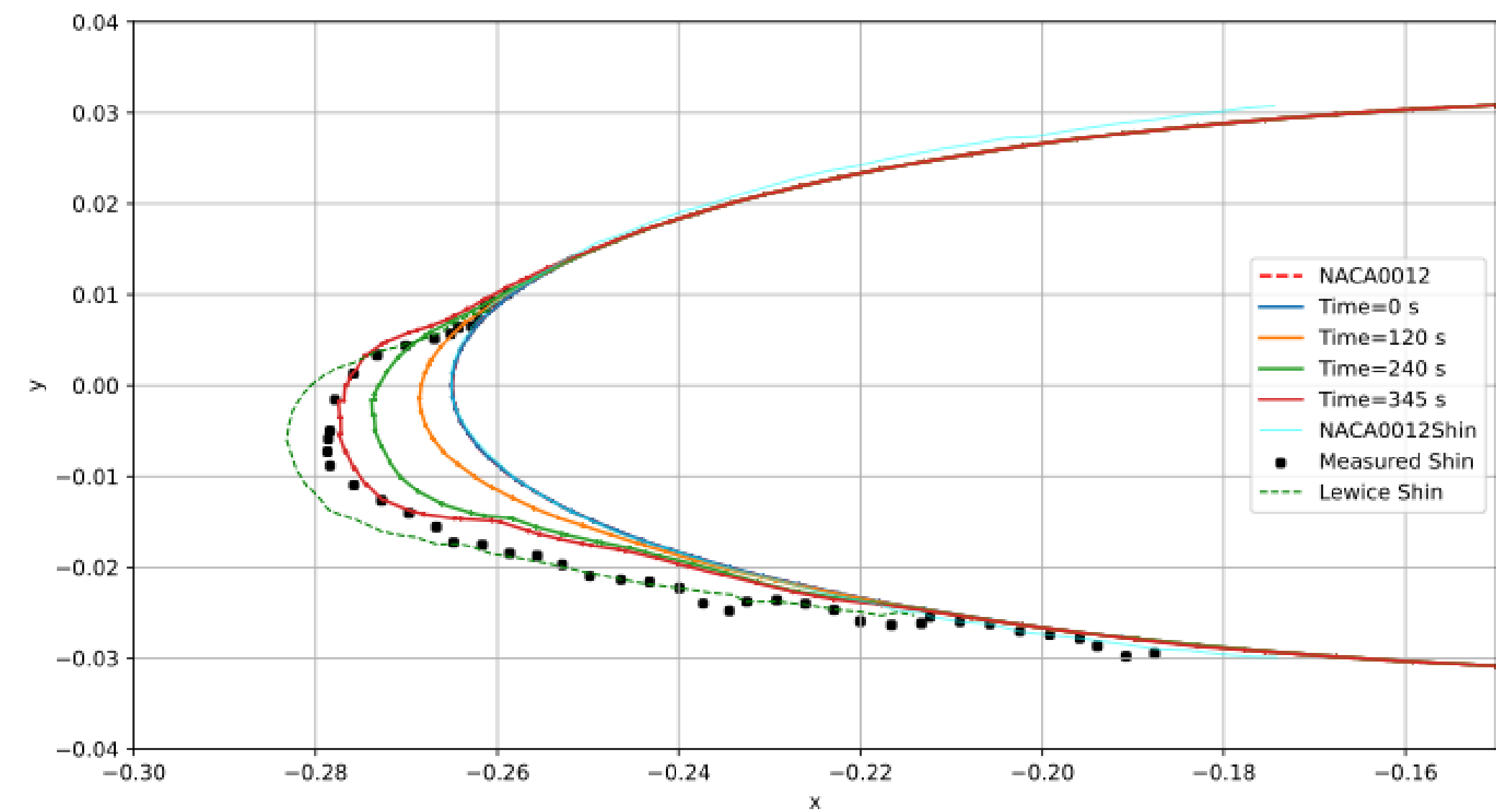
$$h_{\text{ice}} = ff^* \cdot \left(\frac{dh_f}{dt} \right)_{\text{cv, liquid}} \cdot \frac{\rho_{\text{liquid}}}{\rho_{\text{ice}}} \cdot \Delta t_{\text{ice}},$$

Results

Rime Ice Case

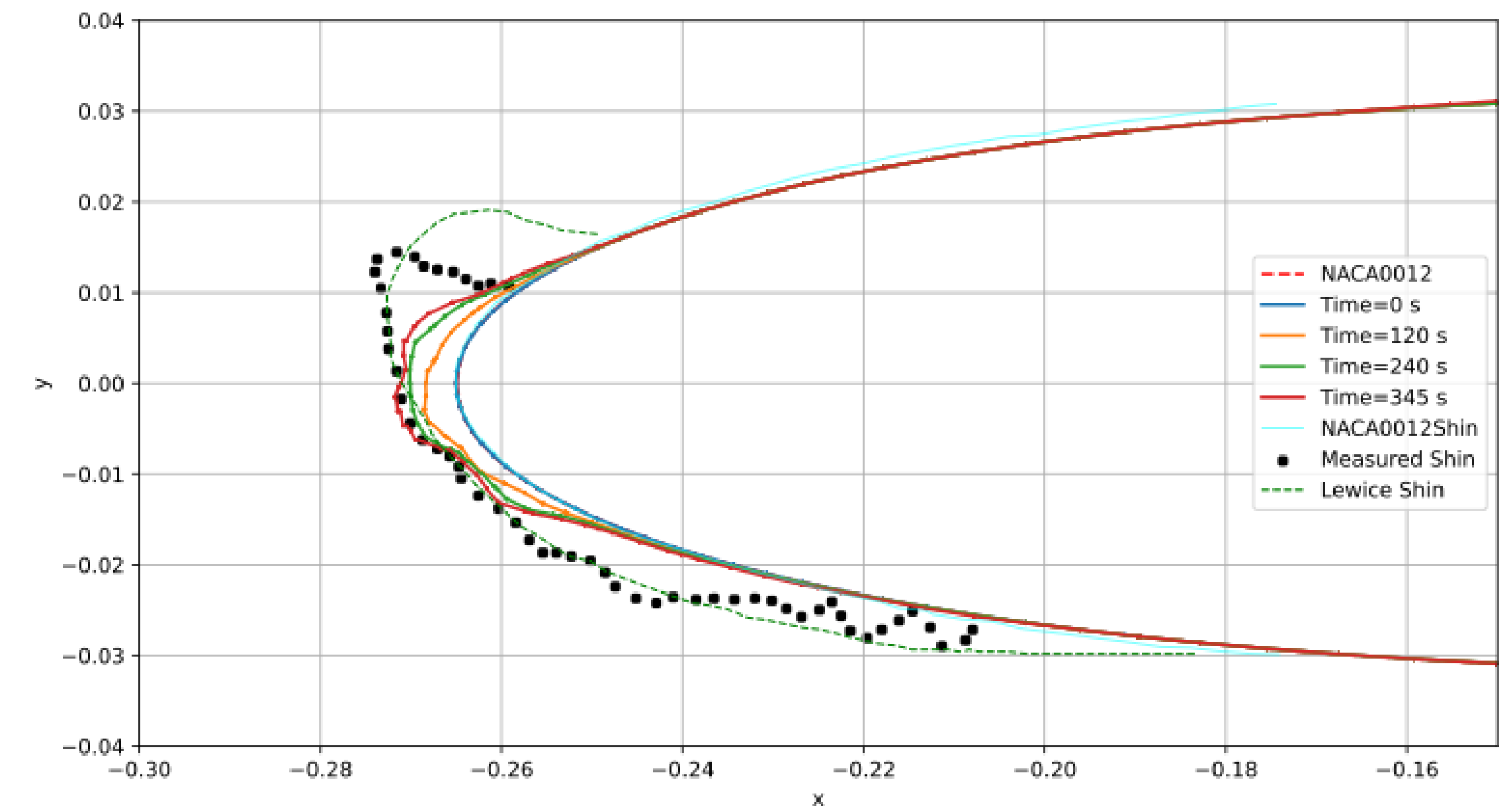
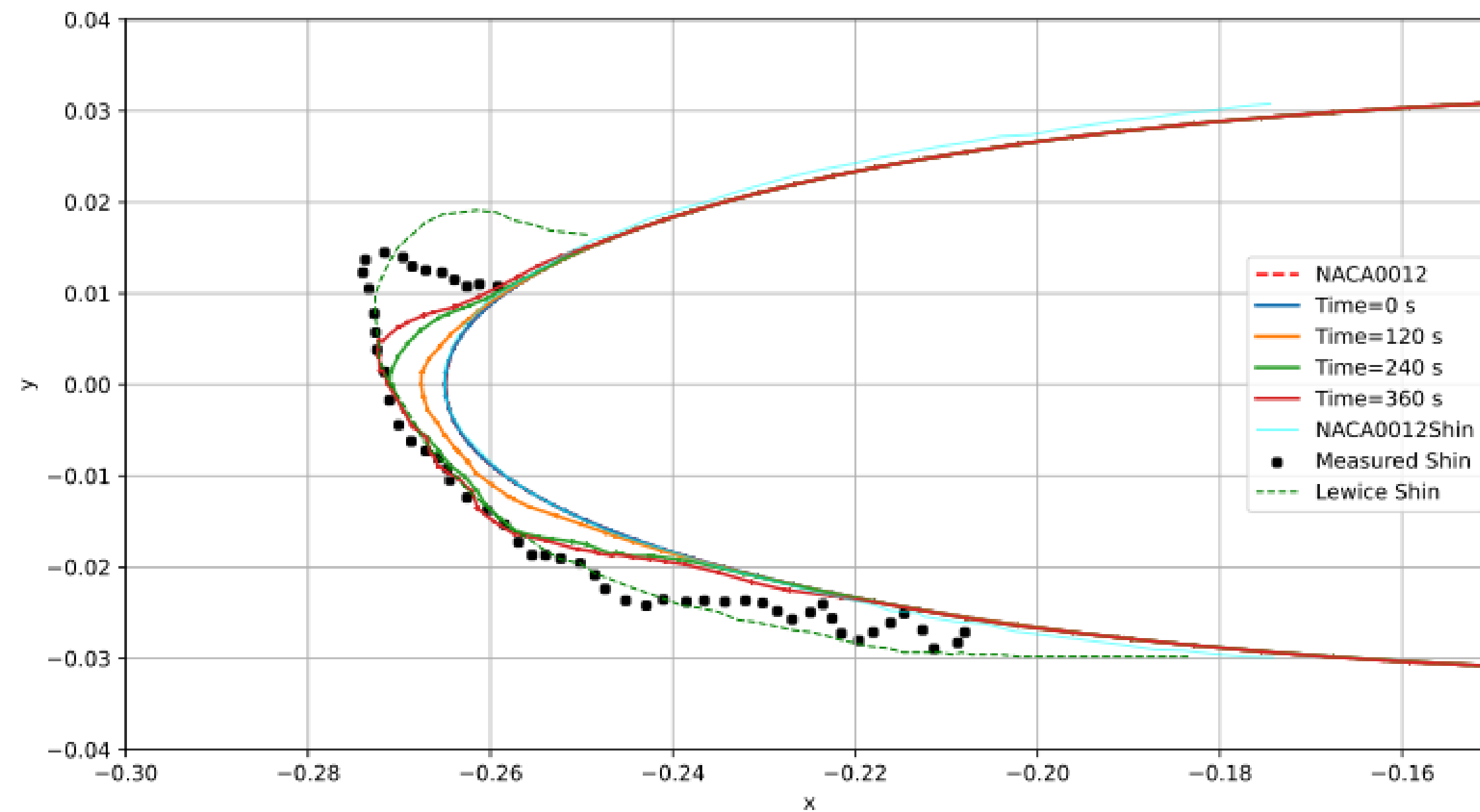


. Transfer function with source normal and cell centres vectors.



. Transfer function with alignment of cell source and target normals.

Glaze Ice Case



. Transfer function with source normal and cell centres vectors.

. Transfer function with alignment of cell source and target normals.

Discussion

- Rime and glaze ice are working with same equations
- Rime is considered acceptable
- Glaze shows a correct behavior but requires adequate:
 - i. thermal freezing fraction - Messinger
 - ii. mapping by cells following the ice growth deformation
 - iii. convective heat and mass transfer with transition

Next Steps

Next Steps

This work has introduced a mature (less risk) development phase of a **rime and glaze 3D ice accretion tool, built on foam- extend 5.0.**

The nexts steps are:

- Beta version release 10 months ahead
- MVP in 2 years with application to real cases

Thank you for your attention

ATS is Global



USA, Europe and Brazil

São Paulo - Brazil, since 2007

Miami - USA, since 2018

São José dos Campos - Brazil, since 2019

Munich - Germany, since 2022

ATS combines high-tech computational methods with traditional engineering analysis to provide accurate solutions for thermal and fluid dynamic systems



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About ATS

The Client Challenges

To have one
contract with ONE
multi-skilled and
experienced company

To mitigate risks by
hiring experience.

To decrease the number of
prototypes and tests.

In addition

To cover peak workloads.



To support the internal team with
missing specific skills.



All skills in a
single company

Engineering,
Simulation, Testing and
Certification



End-to-end Solutions



Turnkey Job

17 years of experience
applying engineering and
CFD in the industry

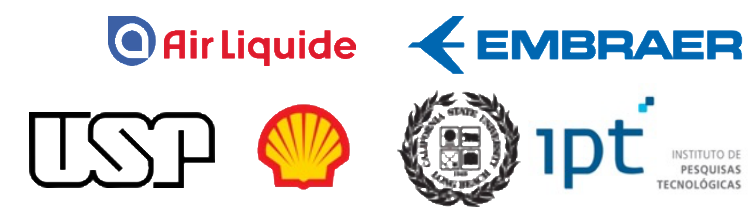
Associate Engineer throughout
the development process

Multi-disciplinary team

Multidisciplinary Team



Guilherme Lima da Silva,
Ph.D, CEO, founder



Pedro Villela
CTO, System Engineer,
Associate



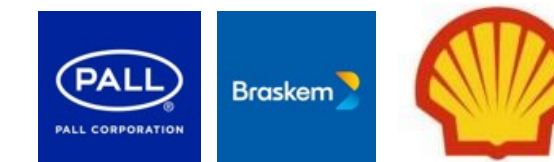
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